

239(5): Key Equations

The relativistic angular momentum is defined by:

$$L = \gamma mr^2 \frac{d\theta}{dt} = mr^2 \frac{d\theta}{d\tau} - (1)$$

in which:

$$L_0 = \frac{L}{\gamma} = \text{constant of motion}. - (2)$$

Therefore:

$$\frac{d\theta}{dt} = \frac{L_0}{mr^2} = \left(\frac{L}{\gamma}\right) \frac{1}{mr^2}. - (3)$$

and

$$t = \int dt = \int \left(\frac{mr^2}{L_0}\right) d\theta - (4)$$

i.e.

$$t = \frac{m}{L_0} \int f^2(\theta) d\theta - (5)$$

Note carefully that L is not a constant, it is defined by:

$$L = L_0 \gamma - (6)$$

where

$$L_0 = mr^2 \frac{d\theta}{dt} = \text{constant}. - (7)$$

The time t in eq. (5) is the time in the

frame of reference in which the planet of mass  $m$  is moving.

The proper time  $\tau$  is the time in which the planet of mass  $m$  is stationary with respect to a frame of reference fixed on the planet. So the time measured on the planet is  $\tau$ .

Here:

$$\gamma = \frac{dt}{d\tau} \quad - (8)$$

so

$$t = \gamma \tau \quad - (9)$$

From eqs. (5) and (9):

$$\tau = \frac{m}{L_0} \int \gamma f^2(\theta) dt \quad - (10)$$

Therefore plot and animate eqs. (5) and (10).

The calculation of the relativistic force of Minkowski proceeds as follows. It is defined as:

$$\underline{F} = m \underline{a} = m \frac{d}{d\tau} \left( \frac{dx}{d\tau} \right) \quad - (11)$$

where:

$$\underline{a} = \frac{d}{d\tau} \left( \frac{dr}{d\tau} \right) = \frac{d}{d\tau} \left( \gamma \frac{dr}{dt} \right) = \gamma \frac{d}{dt} \left( \frac{\gamma dr}{dt} \right) \quad -(12)$$

i.e.  $\underline{a} = \gamma \left( \frac{d\gamma}{dt} \frac{dr}{dt} + \gamma \frac{d}{dt} \left( \frac{dr}{dt} \right) \right) \quad -(13)$

where

$$\gamma = \left( 1 - \frac{v^2}{c^2} \right)^{-1/2} \quad -(14)$$

$$= \frac{dt}{d\tau}.$$

In eq. (14),  $v$  is defined by:

$$v^2 = \left( \frac{dr}{dt} \right)^2 + r^2 \left( \frac{d\theta}{dt} \right)^2 \quad -(15)$$

In eq. (15):

$$\frac{dr}{dt} = \frac{dr}{d\theta} \frac{d\theta}{dt} \quad -(16)$$

so:

$$v^2 = \left( \frac{dr}{d\theta} \right)^2 \left( \frac{d\theta}{dt} \right)^2 + r^2 \left( \frac{d\theta}{dt} \right)^2 \quad -(17)$$

From eq. (3):

$$\left( \frac{d\theta}{dt} \right)^2 = \frac{L_0^2}{m^2 r^4} \quad -(18)$$

So:

$$v^2 = \left(\frac{L_0}{mr^2}\right)^2 \left(r^2 + \left(\frac{dr}{d\theta}\right)^2\right) - (19)$$

For an elliptical orbit:

$$r = \frac{d}{1 + \epsilon \cos \theta} - (20)$$

$$\frac{dr}{dt} = \frac{\epsilon r^2}{d} \sin \theta. - (21)$$

so

$$v^2 = \left(\frac{L_0}{md}\right)^2 \left(1 + \epsilon^2 + 2\epsilon \cos \theta\right) - (22)$$

for an elliptical orbit.

Note carefully that  $L_0$  is fixed in the definition of  $v$  of the Lorentz factor.

Using the chain rule:

$$\frac{dY}{dt} = \frac{dY}{dv} \frac{dv}{dt} - (23)$$

then  $a$  of eqn. (13) becomes:

$$\underline{a} = \gamma^4 \frac{v}{c^2} \frac{dv}{dt} \frac{dr}{dt} + \gamma^3 \frac{d}{dt} \left(\frac{dr}{dt}\right) - (24)$$

5) Therefore:

$$\begin{aligned}\underline{\underline{a}} &= \gamma \frac{d\gamma}{dt} \frac{d\underline{r}}{dt} + \gamma^2 \frac{d}{dt} \left( \frac{d\underline{r}}{dt} \right) - (25) \\ &= \gamma \frac{d\gamma}{dt} \left( \frac{d\underline{r}}{dt} \underline{\underline{e}}_r + \underline{\omega} \times \underline{\underline{r}} \right) \\ &\quad + \gamma^2 \left( \frac{d^2 \underline{r}}{dt^2} \underline{\underline{e}}_r + \underline{\omega} \times (\underline{\omega} \times \underline{\underline{r}}) \right)\end{aligned}$$

Because the Coriolis acceleration is zero for all planar orbits.

It follows that:

$$\begin{aligned}\underline{\underline{a}} &= \left( \gamma^2 \frac{d^2 \underline{r}}{dt^2} + \gamma \frac{d\gamma}{dt} \frac{d\underline{r}}{dt} \right) \underline{\underline{e}}_r \\ &\quad + \gamma^2 \underline{\omega} \times (\underline{\omega} \times \underline{\underline{r}}) + \gamma \frac{d\gamma}{dt} \underline{\omega} \times \underline{\underline{r}}\end{aligned} \quad -(26)$$

Using eq. (23):

$$\frac{d\gamma}{dt} = \frac{\gamma^3 v}{c^2} \frac{dv}{dt}, \quad -(27)$$

and  $v = \frac{d\underline{r}}{dt}, \quad \frac{dv}{dt} = \frac{d^2 \underline{r}}{dt^2} \quad -(28)$

So:

$$\underline{a} = \gamma^4 \frac{d^2 r}{dt^2} \underline{e}_r + \gamma^3 \underline{\omega} \times (\underline{\omega} \times \underline{r}) + \gamma \frac{d\gamma}{dt} \underline{\omega} \times \underline{r} \quad -(29)$$

Now use:

$$\underline{\omega} \times (\underline{\omega} \times \underline{r}) = - \frac{L_0^2}{m^2 r^3} \underline{e}_r \quad -(30)$$

and

$$\underline{\omega} \times \underline{r} = \omega r \underline{e}_\theta \quad -(31)$$

To find that:

$$\underline{a} = \left( \gamma^4 \frac{d^2 r}{dt^2} - \frac{\gamma^2 L_0^2}{m^2 r^3} \right) \underline{e}_r + \frac{\gamma^4}{c^2} \frac{dr}{dt} \frac{d^2 r}{dt^2} \omega r \underline{e}_\theta \quad -(32)$$

where

$$\underline{e}_r = \cos\theta \underline{i} + \sin\theta \underline{j} \quad -(33)$$

$$\underline{e}_\theta = -\sin\theta \underline{i} + \cos\theta \underline{j} \quad -(34)$$

$$\gamma = \left( 1 - \frac{v^2}{c^2} \right)^{-1/2} \quad -(35)$$

$$v^2 = \left( \frac{L_0}{mr} \right)^2 \left( r^2 + \left( \frac{dr}{d\theta} \right)^2 \right) \quad -(36)$$